

## SUPPLEMENTAL TYPE CERTIFICATE

## 10054584

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

# Q.C.M. DESIGN GmbH

EICHHOLZWEG 20-24 3123 BELP SWITZERLAND

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

**Type:** AS 350/EC 130 **Model:** EC 130 T2

Description of Design Change: Hook Release Modification

Installation of a switch to allow the pilot the option of selecting either the existing upper hook or a lower hook for cargo release.

#### **EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 27 August 2015

Massimo MAZZOLETTI

**Head of Rotorcraft Department** 

10036991

SUPPLEMENTAL TYPE CERTIFICATE - 10054584 - Q.C.M. DESIGN GmbH - 305971





### **Associated Technical Documentation:**

E350-QCM25-001-M1-HFM-R1 rev. one. Flight Manual Supplement

or later revisions of the above listed documents approved by EASA.

CP-E350-QCM25-001 rev. 2. Certification Program

CR-E350-QCM25-001 rev. 1. Compliance Report

E350-QCM25-001-M1-ICA initial issue. Instruction for Continued Airworthiness

E350-QCM25-001-M1-MMS initial issue. Maintenance Manual Supplement

### **Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

