SUPPLEMENTAL TYPE CERTIFICATE

10053613

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

Q.C.M. DESIGN GmbH

EICHHOLZWEG 20-24
3123 BELP
SWITZERLAND

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350/EC 130

Model: EC 130 T2

Description of Design Change:
E350-QCM34-001 Installation of an Avidyne TAS605 TAS
Removed: - N/A
Installed: - Avidyne TAS 605 Traffic Advisory System (TAS)

EASA Certification Basis:
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 10 June 2015

Massimo MAZZOLETTI
Head of Rotorcraft Department

Associated Technical Documentation:
Flight Manual Supplement: E350-QCM34-001-M1-HFM-R2, dated 20.05.2015
or later revisions of the above listed documents approved by EASA.
Master Document List: E350-QCM34-001-R1_Section 2, dated 10.06.2015
Instructions for continued airworthiness: -E350-QCM34-001-M1-ICA-R0, dated 16.03.2015

Limitations:
Limited to S/N 7989

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -