

SUPPLEMENTAL TYPE CERTIFICATE

10072321

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

Q.C.M. DESIGN GmbH

EICHHOLZWEG 20-24
3123 BELP
SWITZERLAND
EASA.21J.509

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.155

Type Certificate Holder: DASSAULT AVIATION

Type: Falcon 7X

Model: Falcon 7X

Description of Design Change:

Removal of A/C Cabin & Lav Ashtrays

As smoking is not permitted on the aircraft, all the ashtrays currently installed on the passenger cabin and lavatory are removed.

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment: CS-25 25.1; 25.21; 25.23; 25.25; 25.27; 25.29; 25.791; 25.853; 25.854; 25.1529; 25.1541; 25.1581; CS 25 Appendix H H25.1; H25.2; H25.3; H25.4; H25.5; CS 25 Appendix S S25.1; S25.10; S25.20; S25.30; S25.40; S25.50 at amdt. 19.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 30 January 2020



Dominique RENAUD
Section Manager
Jet Aeroplanes



Associated Technical Documentation:

- AFM DF8X-QCM25-001-M1-AFM Rev. 00
- MDL DF8X-QCM25-001 Rev. 00
- ICA DF8X-QCM25-001-M1-ICA

or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

