

SUPPLEMENTAL TYPE CERTIFICATE

10057994

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

Q.C.M. DESIGN GmbH

**EICHHOLZWEG 20-24
3123 BELP
SWITZERLAND**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.078

Type Certificate Holder: TEXTRON AVIATION INC.

Type: 525 SERIES

Model: 525

Description of Design Change:

ADS-B Out Upgrade for a Cessna 525

Removed: Garmin GTX-330D Transponder

Installed: Garmin GTX-330D w/ES Transponder

Compliance to CS-ACNS.D.ADSB and AMC 20-24 was shown.

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 04 May 2016



Yves MORIER

**Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)**

10043321
SUPPLEMENTAL TYPE CERTIFICATE - 10057994 - Q.C.M. DESIGN GmbH - 305971

CS-ACNS.D.ADSB

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

- Airplane Flight Manual Supplement: Doc. No. C525-QCM34-001-M1-AFM, Rev. 0
- Instructions for continued Airworthiness: Doc. No. C525-QCM34-001-M1-ICA, Rev. 0
or later revisions of the above listed documents approved by EASA
- Master Document List: Doc. No. Section 2.1 of C525-QCM34-001, Rev. 0
- Declaration of compliance: Doc. No. C525-QCM34-001-M0-DOC Rev. 0
- Engineering Bulletin: Doc. No. EB-C525-QCM34-001 Rev.0

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -