

SUPPLEMENTAL TYPE CERTIFICATE

10064455

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

Q.C.M. DESIGN GmbH

**EICHHOLZWEG 20-24
3123 BELP
SWITZERLAND**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.085

Type Certificate Holder: HAWKER BEECHCRAFT CORPORATION

Type: Hawker Series

Model: Hawker 800XP

Description of Design Change:

Delta Certification of a GPS/WAAS Sensor & Antenna Installation for the support of the Approved EASA STC 10058529.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Installation and Maintenance in accordance with Master Document List RH80-QCM34-001-M0-MDL, Initial Issue, dated 27th November 2017,
or later revisions of the above listed documents approved by EASA.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 23 January 2018



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)

Limitations/Conditions:

- Limited to aircraft serial number 258536.
- The EASA STC 10058529 is a prerequisite for the installation of this STC.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -